





- Very few accidents have occurred where there was loss of normal flight control
- Modern flight control systems are exceptionally well designed
 - Redundancy in electric and hydraulic systems
 - Failure analysis assures 10⁻⁹ reliability
- If loss of normal flight control does occur:
 - Pilot techniques are available to regain control
 - Design features have been identified to allow propulsion control through the autopilot

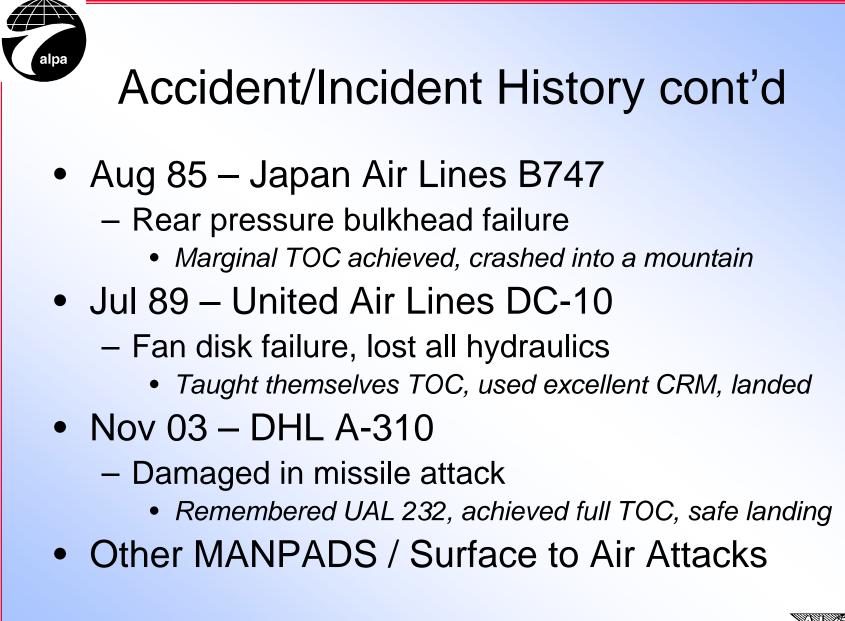


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Accident/Incident History with Loss of Normal Flight Control

- Mar 74 Turkish Airlines DC-10
 - Rear cargo door failure
 - No knowledge of TOC, crashed at high speed
- Apr 75 USAF C-5A "Operation Babylift"
 - Rear pressure door failure
 - Used TOC + wing controls for 30 minutes, crashed
- Apr 77 Delta Airlines L-1011
 - One elevator jammed full-up
 - Used bank angle, then TOC and load shift to gain control









Safety Perspective

- 600M flights from 1974 to 2004
- Total commercial events: 5
 - Rate: .0083 / 1M flights
 - Current Accident Rate .67 / 1M flights
- Total lives lost: 1098
- Of 3447 airplanes among U.S. carriers, 1607, or 46.6% have no mechanical back-up flight controls



Summary of NASA Research

- Tested 7 different airplanes in flight
 - Transports: B747, B777, MD11
 - Fighter/trainer: F-15, F/A-18, T-38
 - Propeller: PA-30
- Results:
 - Gross flight path and heading control possible in all types tested
 - Safe runway landings exceedingly difficult





Achieving Throttles-Only Control

- Motions are affected by trim position, center of gravity, and fuel slosh
- Thrust alone must be used to return to straight and level flight
- Two motions present with loss of normal flight controls:
 - Long period, or *Phugoid*
 - Lateral/directional, or Dutch Roll



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Solve the lateral/directional Problem

- Use asymmetric thrust to return to straight flight:
 - Add thrust to generate sideslip, which generates roll rate
 - Thrust increase lags thrust lever inputs
 - Sideslip lags thrust increase
 - Roll rate lags sideslip
 - Slow inputs avoid fuel slosh
 - In straight flight, some oscillations may remain

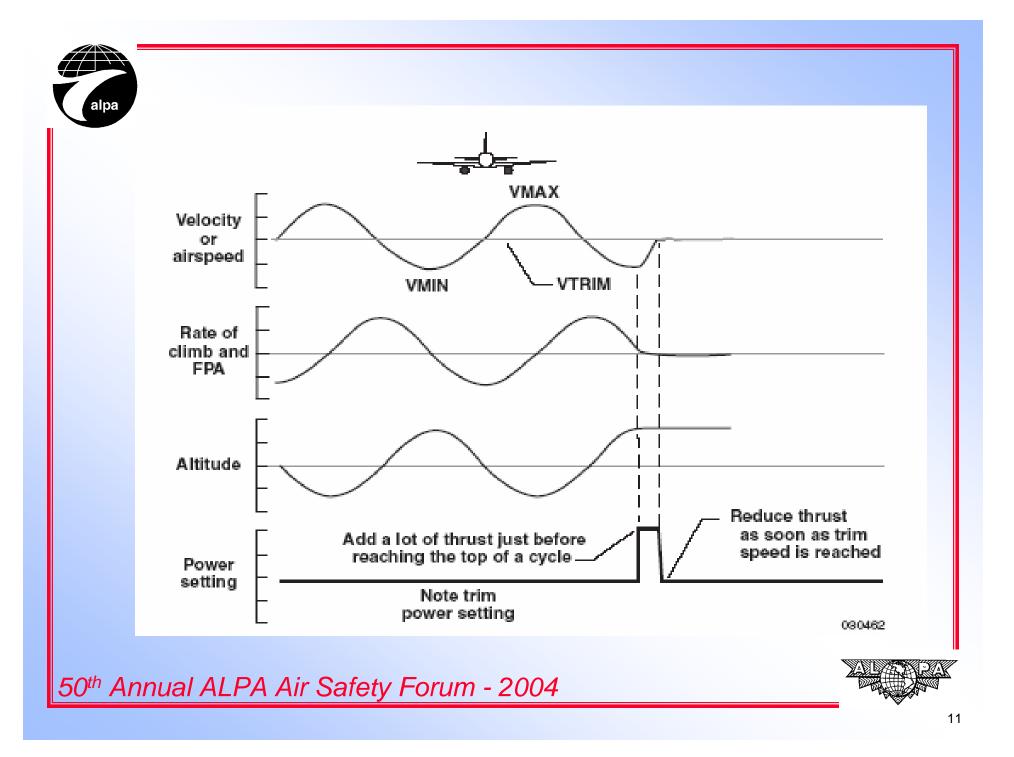




Then Solve the Pitch Problem

- Phugoid oscillations may be 1min or longer
- Avg the high/low speeds to get trim speed
- Add thrust with speed decreasing and the nose near level flight
- Decrease thrust with speed increasing and the nose near level flight
- Continue until oscillations cease, then aggressively maintain airspeed +/- 2 kts







- Pitch and roll are now both under control
- Now, you can think
 - Fly the airplane, and use CRM to divide up all the tasks you face
 - Look at all the EICAS or ECAM messages and determine the exact status of the airplane
 - Use good systems knowledge to determine what you available for additional flight control
 - Begin thinking about where you want to land
 - Trim speed will change with fuel burn or configuration change
 - Make any changes *slowly and incrementally*



Ten Steps to a Survivable Landing Using Only Throttles

NASA's suggested techniques (contained in its report on TOC) have been summarized in the (condensed) list below:

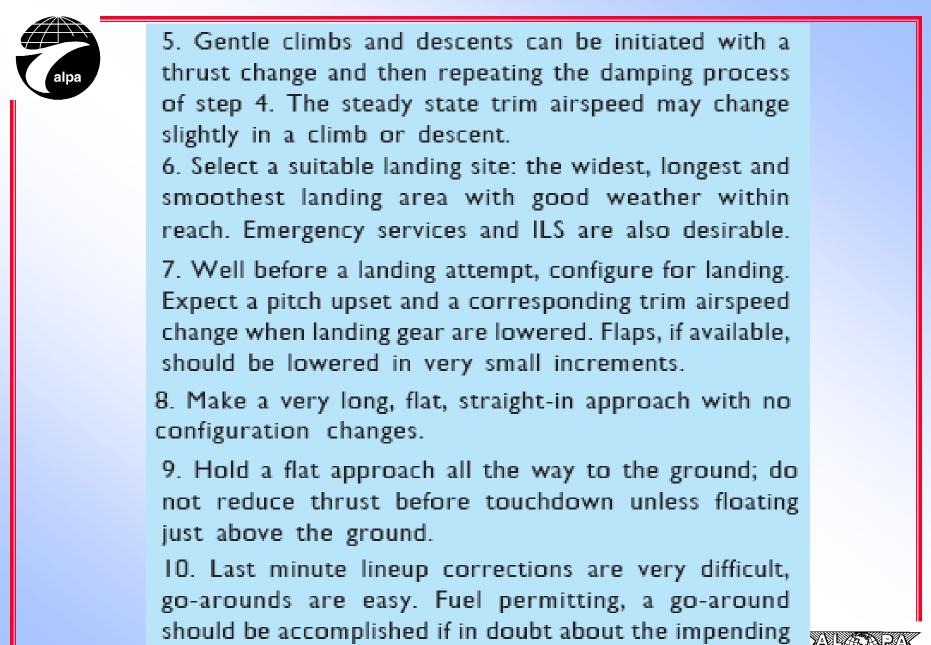
 If a wing is low, push that wing's throttle(s) up until wings are level. Continue to use asymmetric thrust as required to control bank angle and heading.

2. If the pitch attitude and airspeed continually oscillate, determine the approximate steady state trim airspeed by averaging the high and low speeds seen and set a reference bug or mark at that speed.

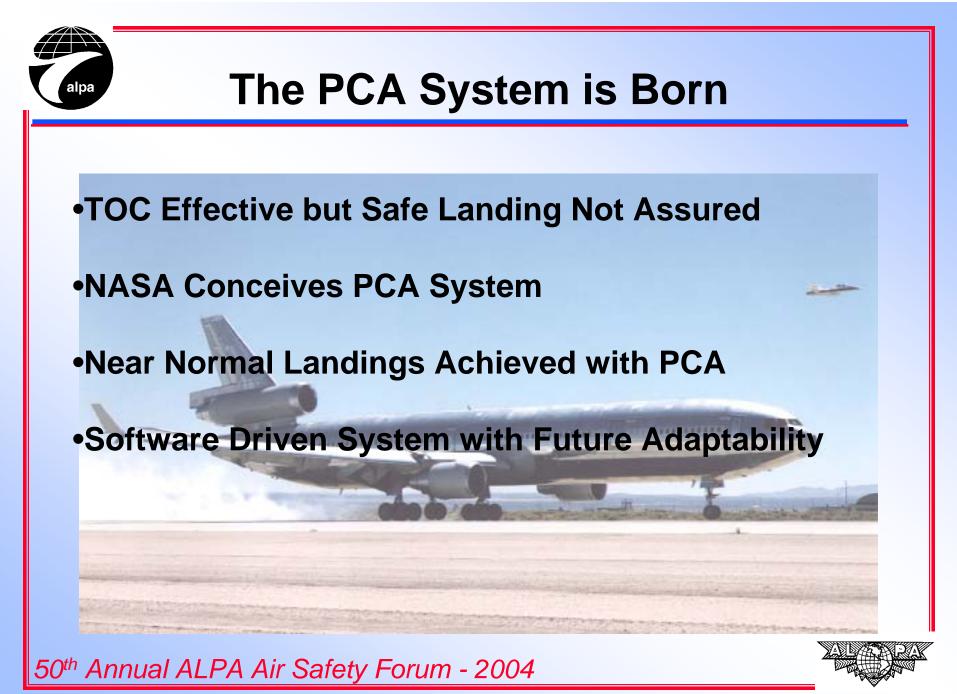
3. Damp the pitch oscillation using aggressive throttle inputs to force the airspeed to the steady state trim airspeed as the nose approaches a level attitude.

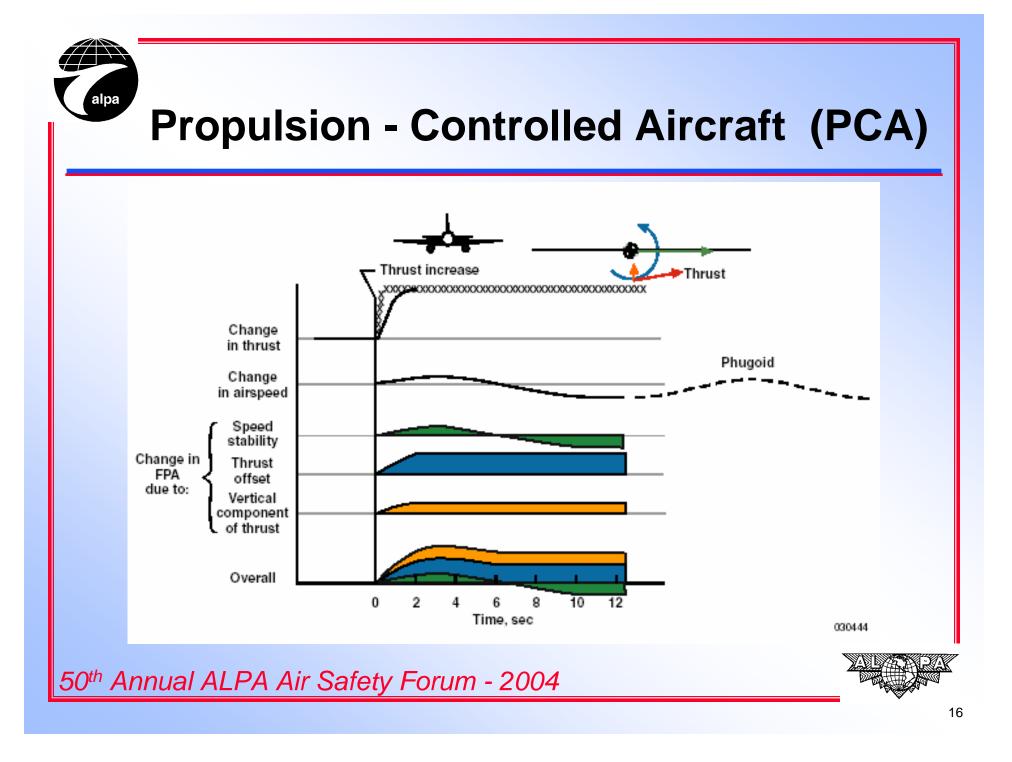
4. Continue this process until all pitch oscillations are stopped. Constant, precise control of airspeed is the key to prevent oscillations from beginning anew.

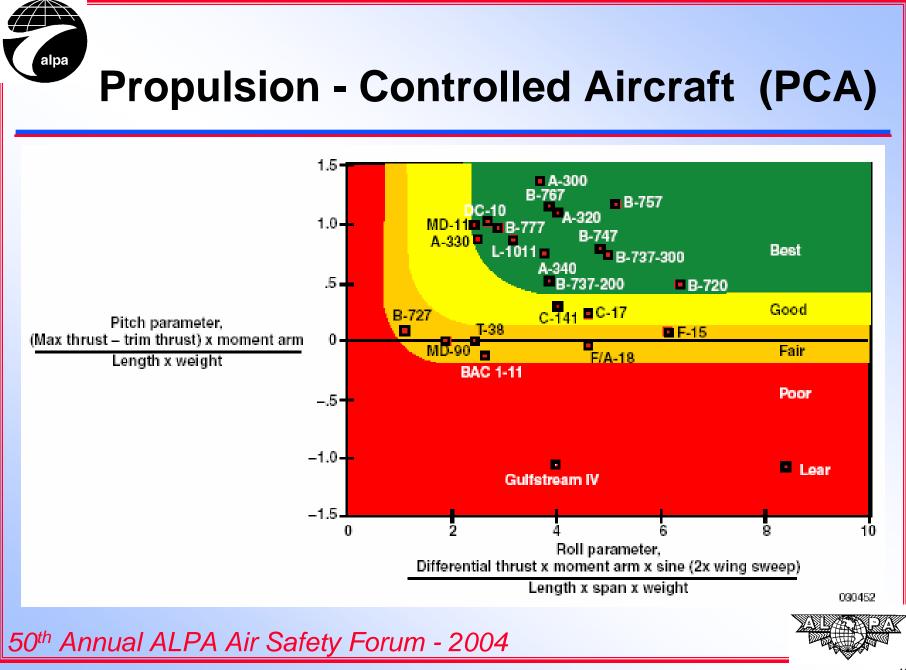


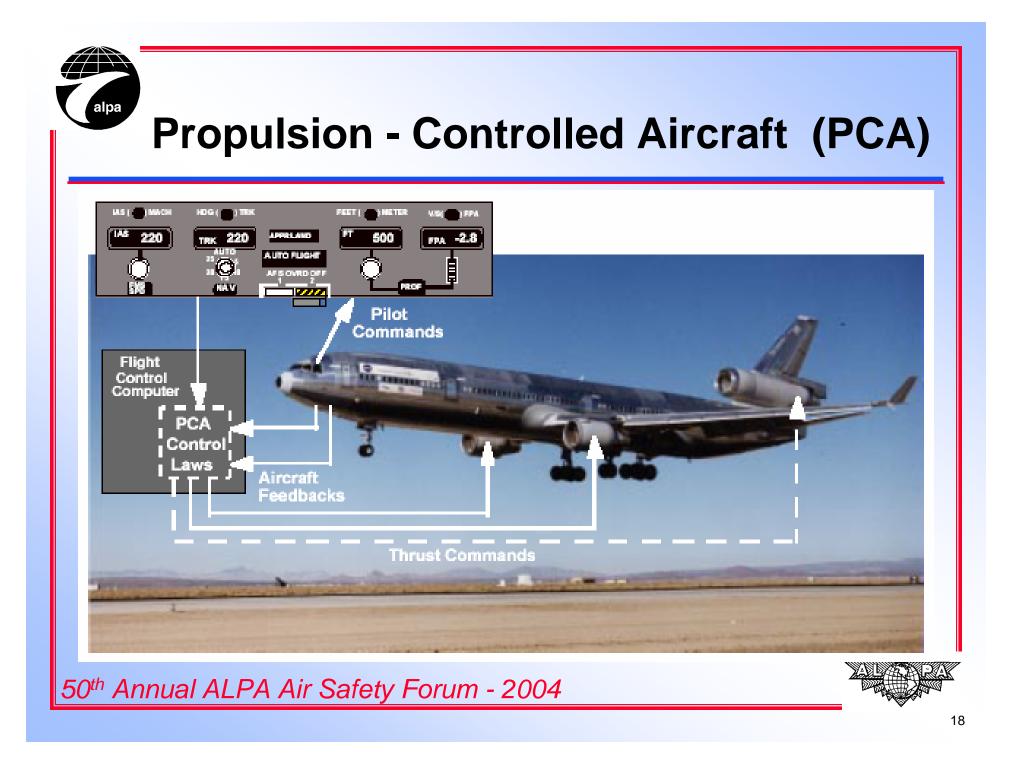


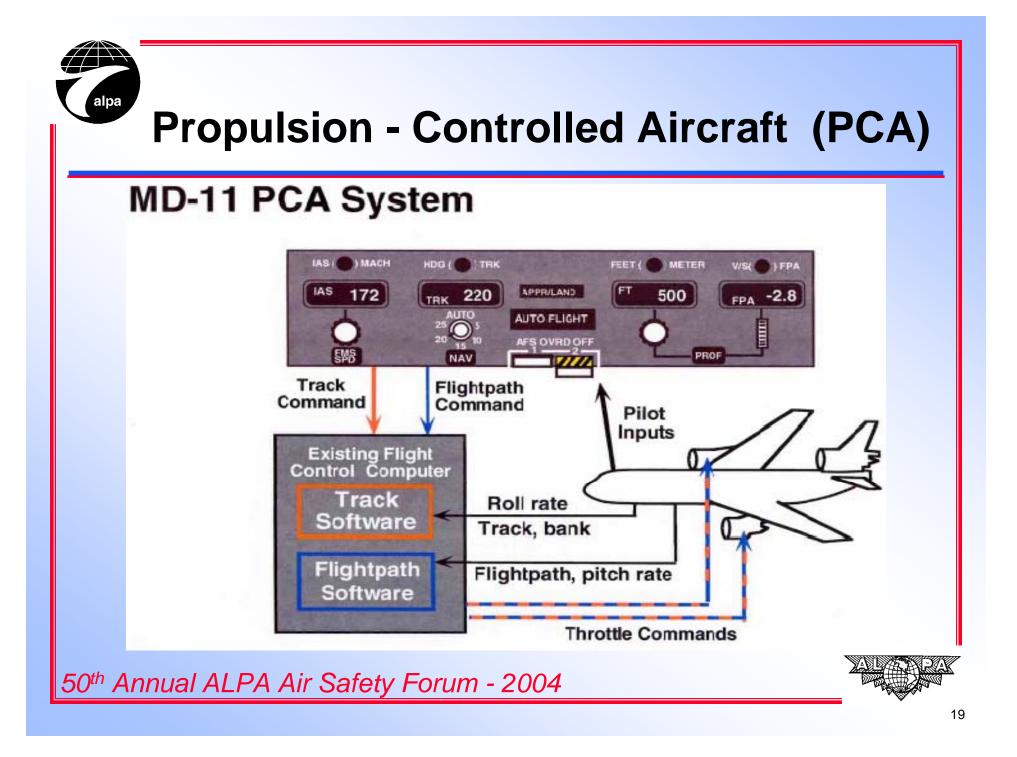
Annua touchdown.

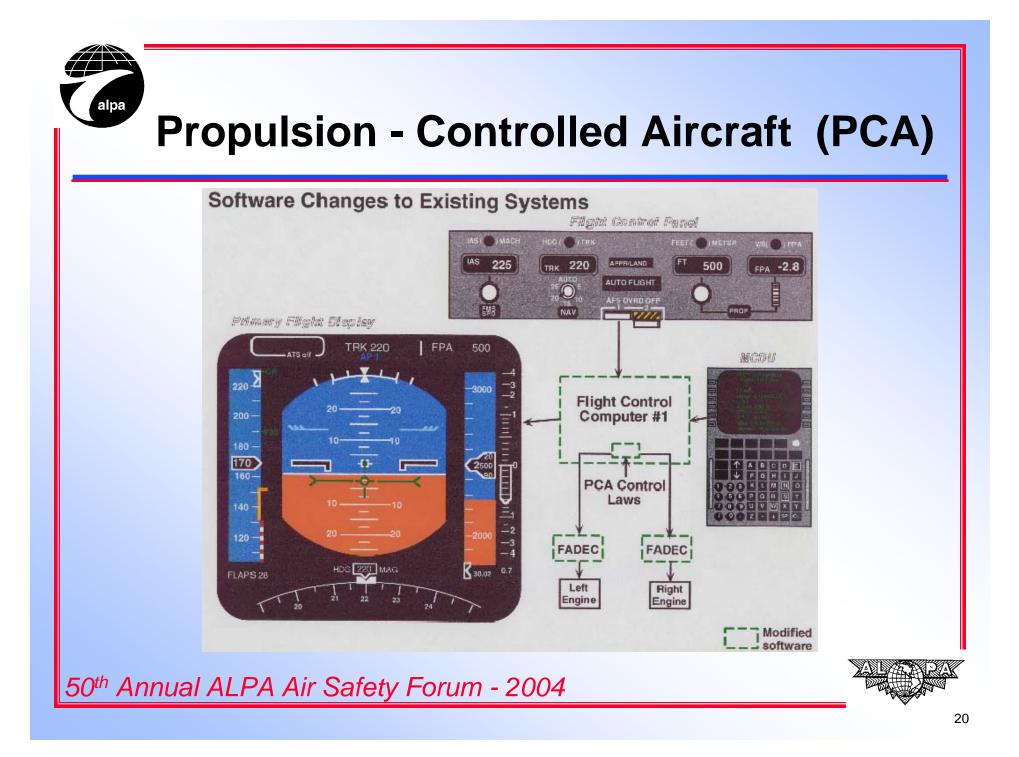


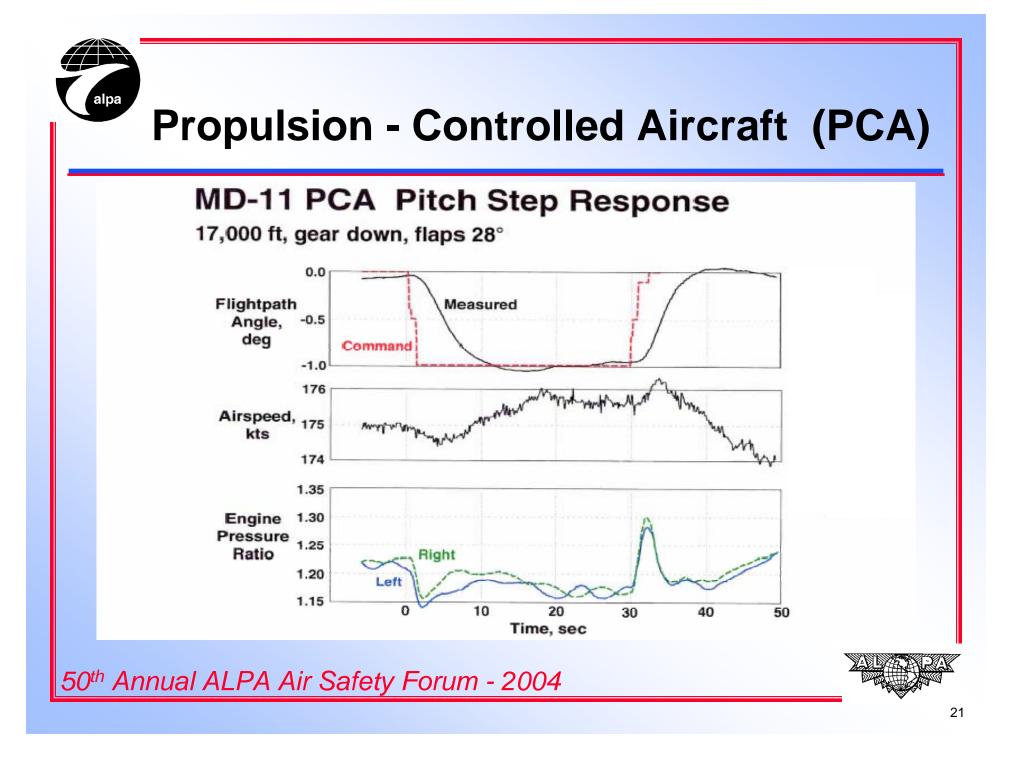


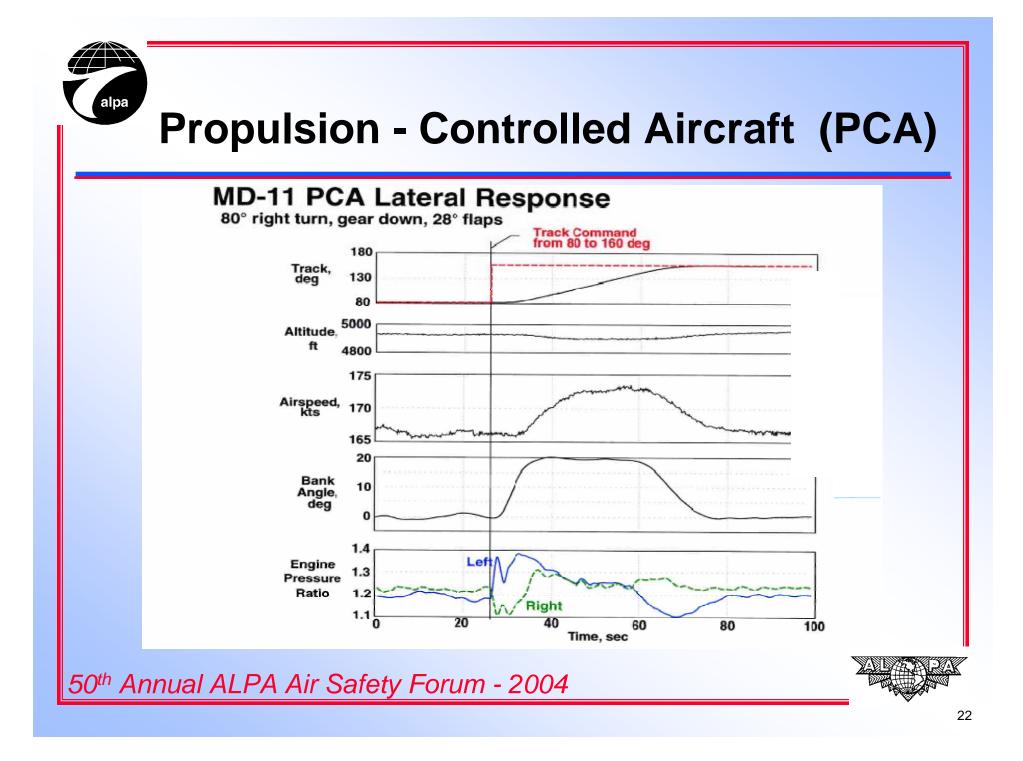


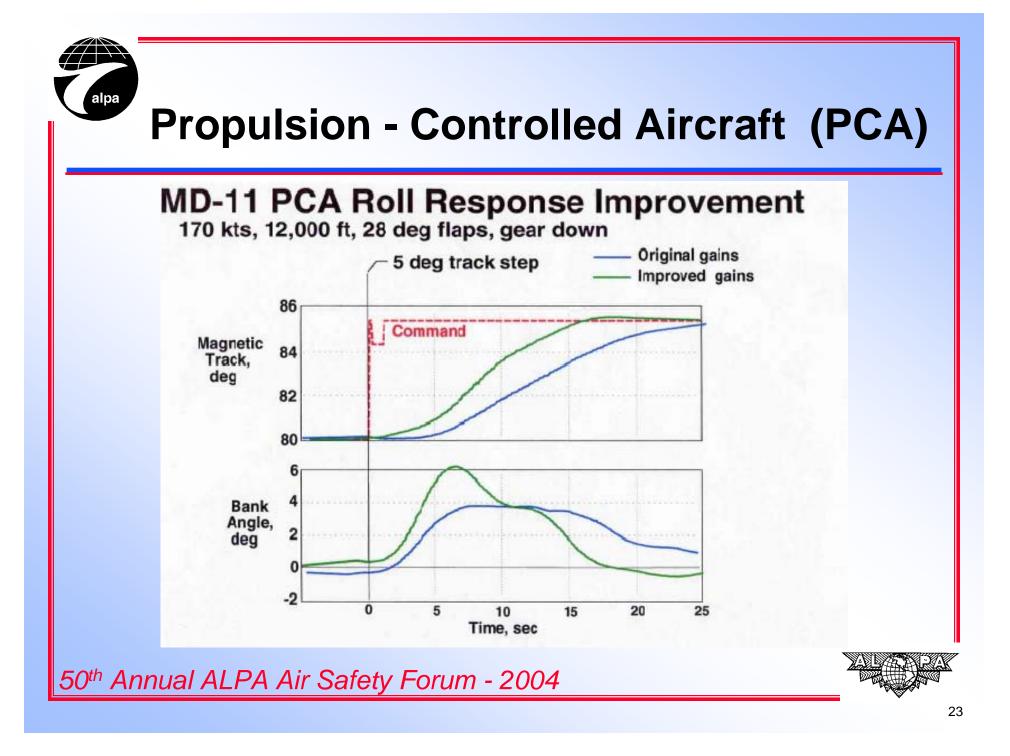




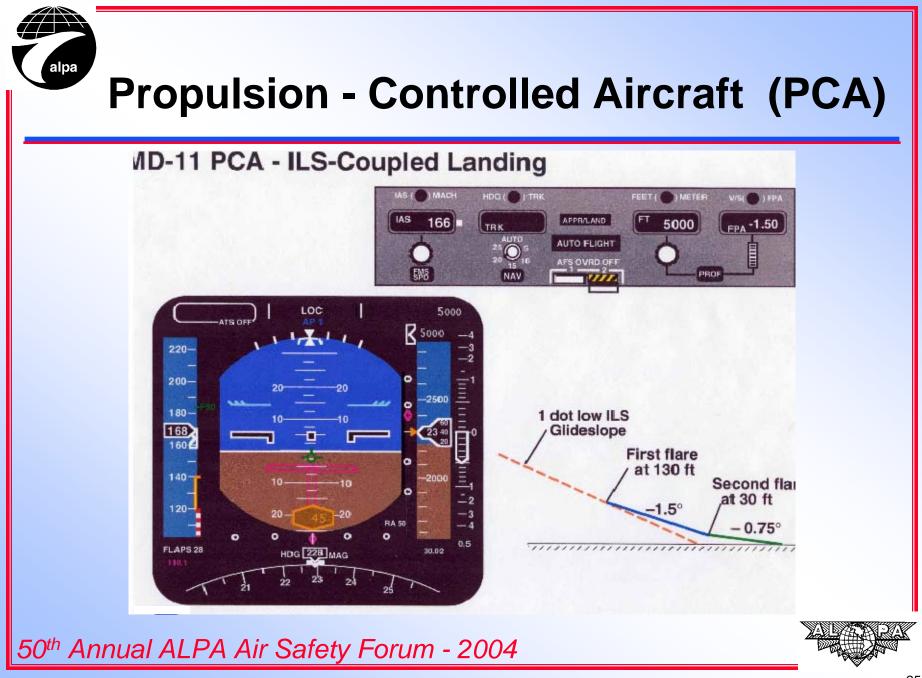


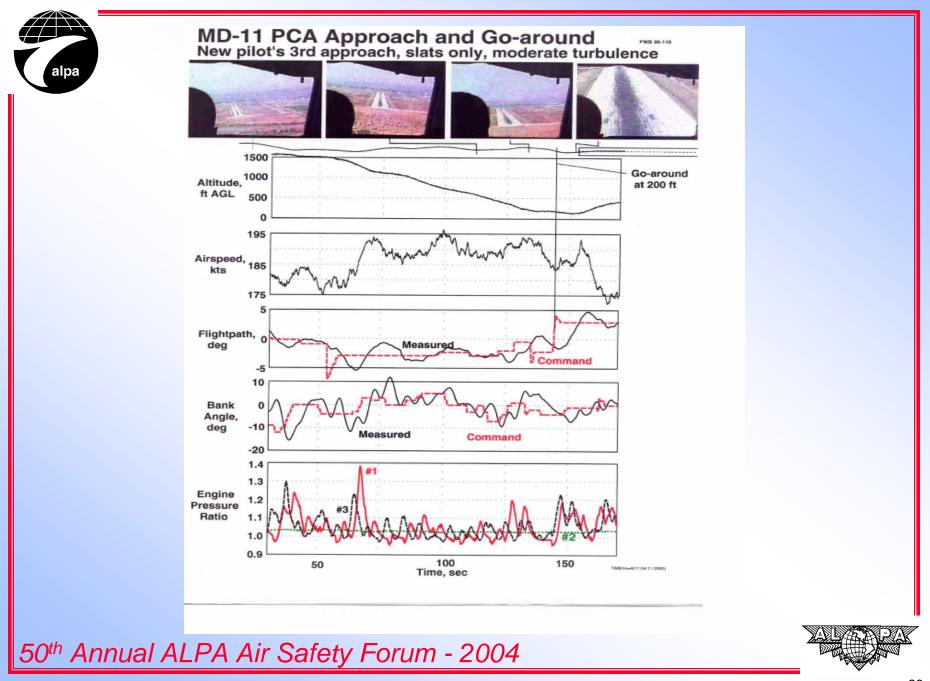


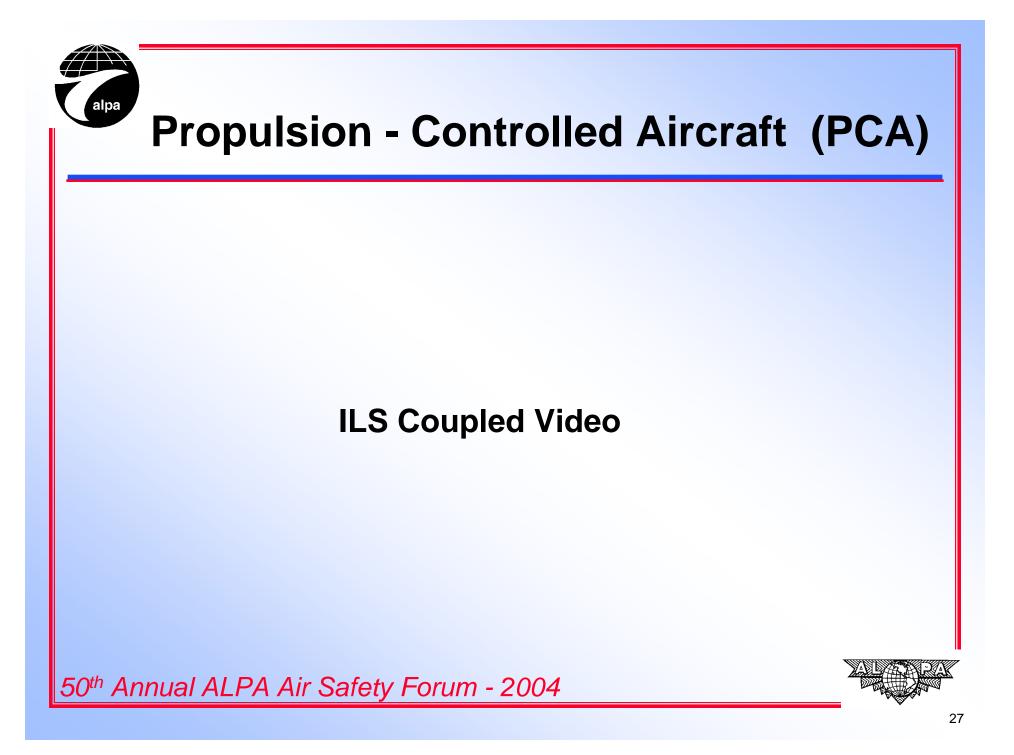


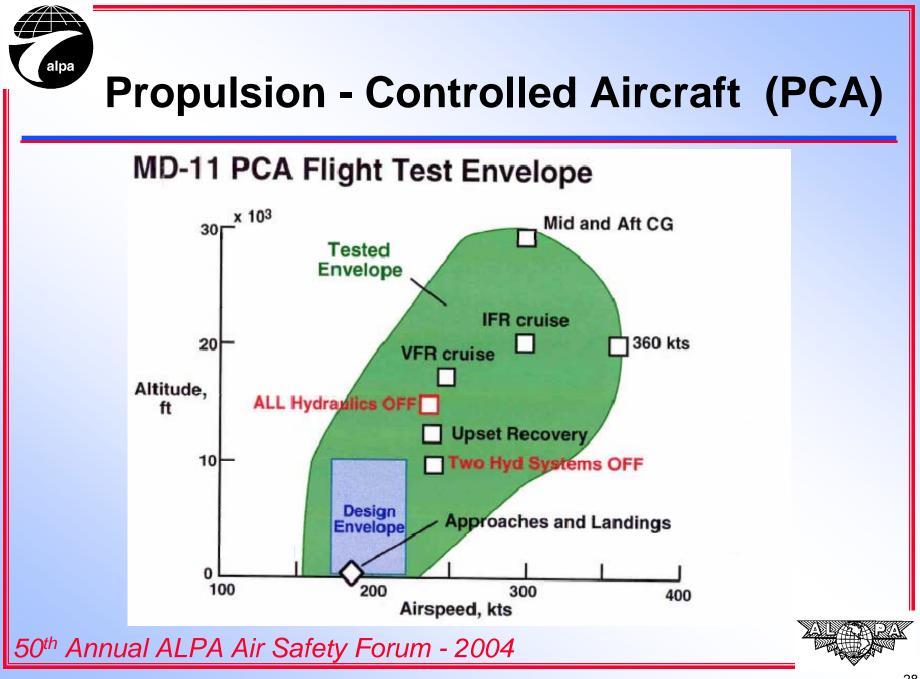












Propulsion - Controlled Aircraft (PCA)

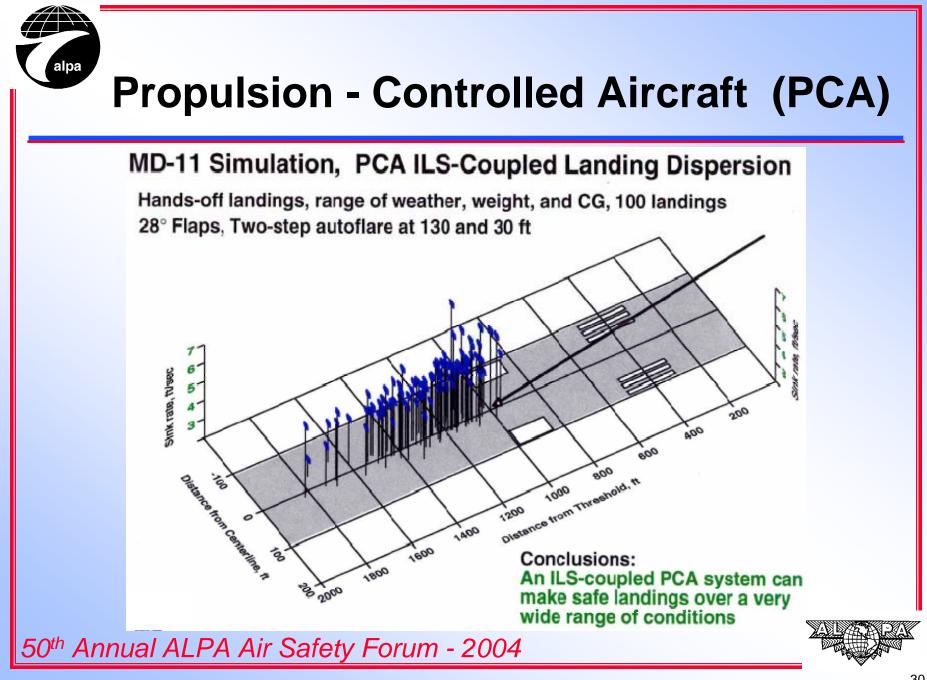
All Hydraulics Off Test

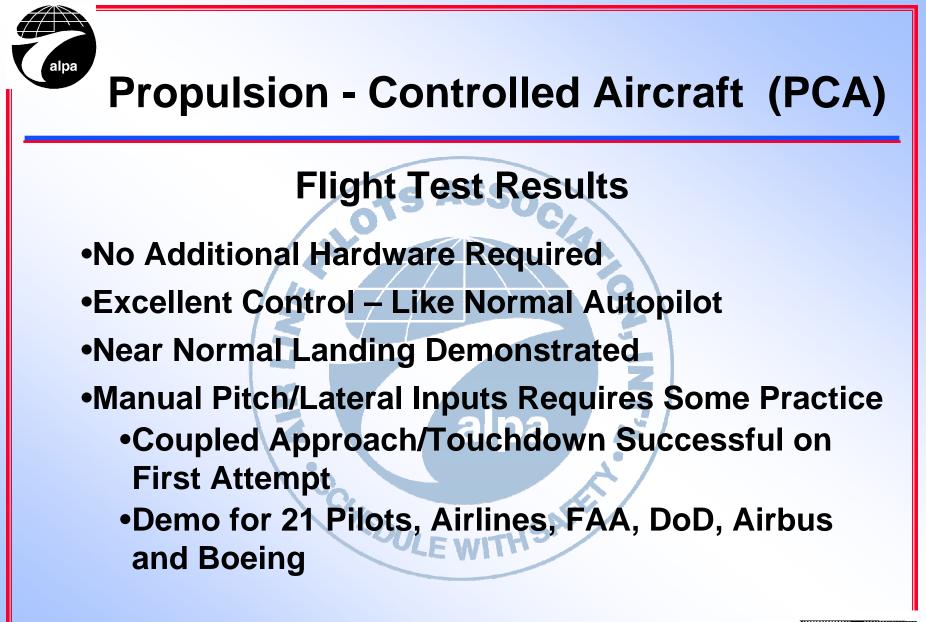


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- Stabilizer remains fixed
- Rudder, elevator float near trim position
- Ailerons float up, mild nose-up trim change
- Alternate gear extension, small nose up trim change
- Flew 25 minutes, PCA control like normal autopilot

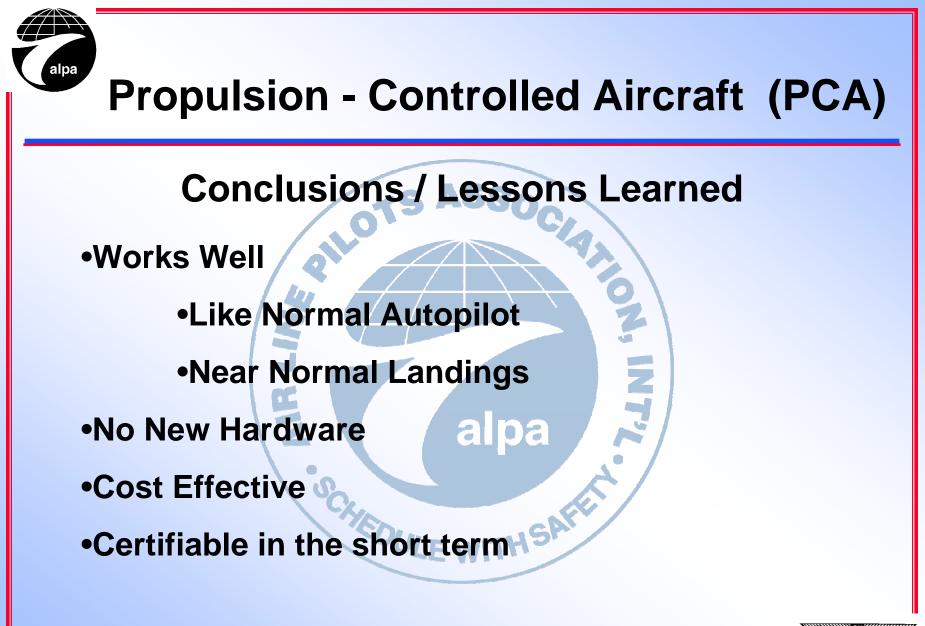


















Throttle Only Control (TOC) Propulsion - Controlled Aircraft (PCA)

Questions?

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Gordon Fullerton – Chief Pilot NASA DFRC

NASA TM-2004-212045 Manual Manipulation of Engine Throttles for Emergency Flight Control

NASA TP-97-206217 Development and Flight Test of an Emergency Flight Control System Using Only Engine Thrust on an MD-11 Transport Airplane

NASA reports www.ifalpa.intranets.com/ members only section

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